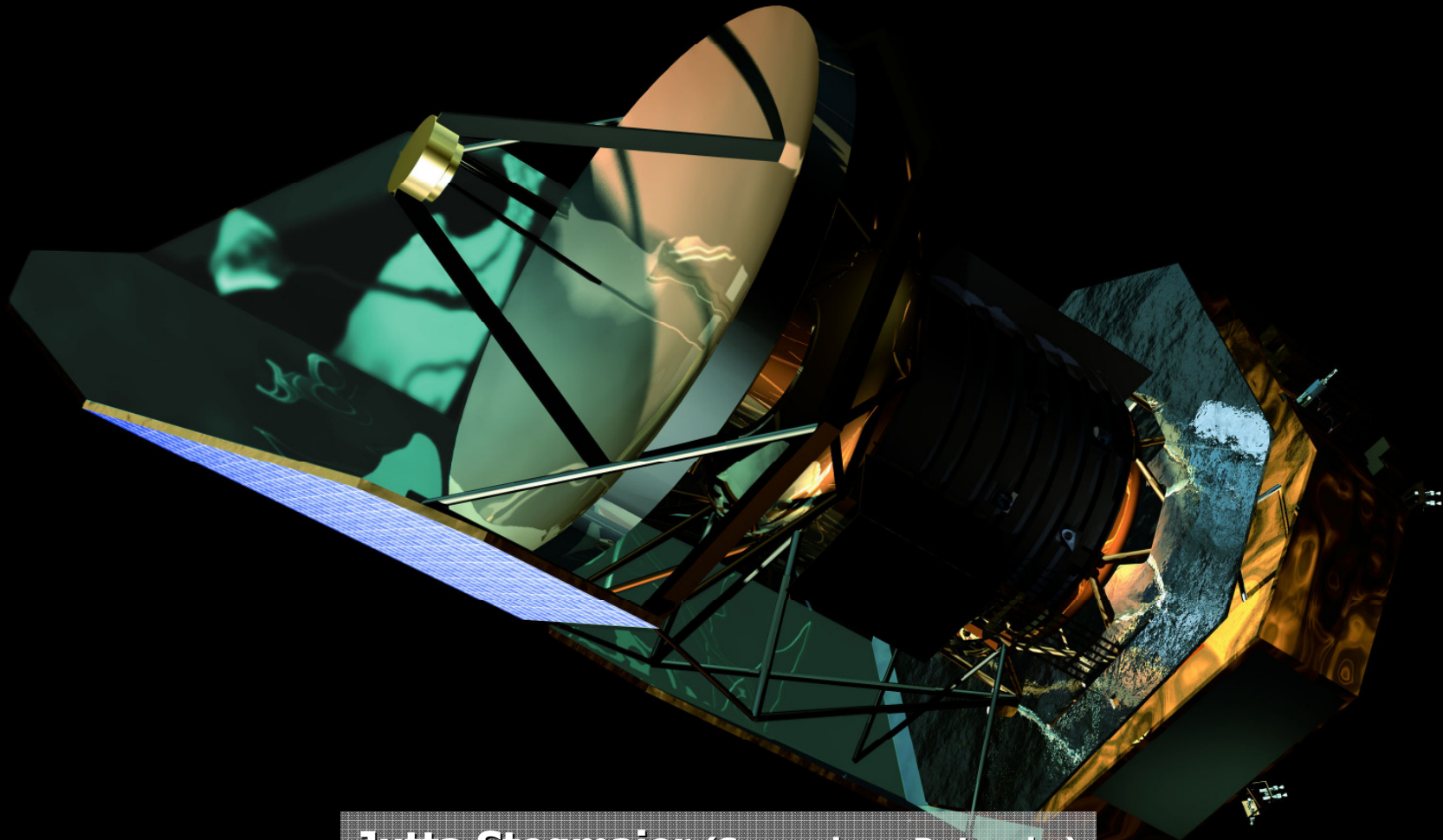


Current Transients in Photodetectors



Jutta Stegmaier (Supervisor: D. Lemke)
4th MPIA Student Workshop
March 26th 2007

What I'm doing

HERSCHEL - PACS

- ✘ Characterisation of Ge:Ga Detectors
- ✘ Laboratory Simulation of Cosmic Radiation at L2 & Curing of Ge:Ga Detectors
- ✘ Moon and beyond

 3rd student workshop

What I'm doing

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- ⌘ Moon and beyond

HERSCHEL - In a Nutshell

Herschel

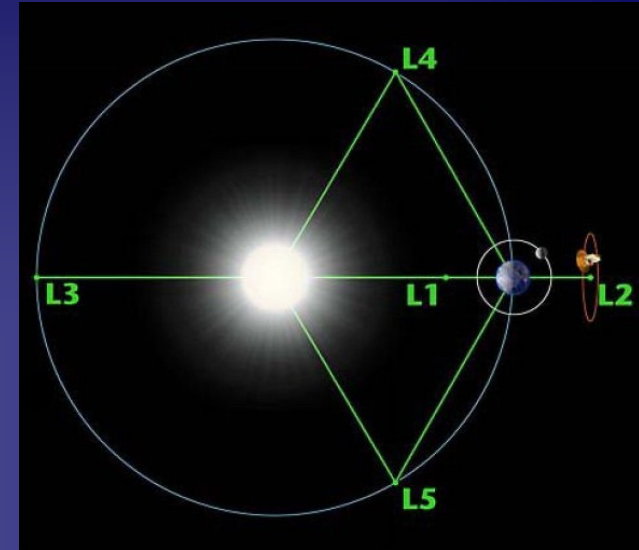
Mirror: 3.5 m, $T \sim 80$ K

Launch: 2008

Orbit around L2

FIR space facility (57-670 μm)

Instruments: PACS, SPIRE, HIFI



HERSCHEL - PACS

Herschel

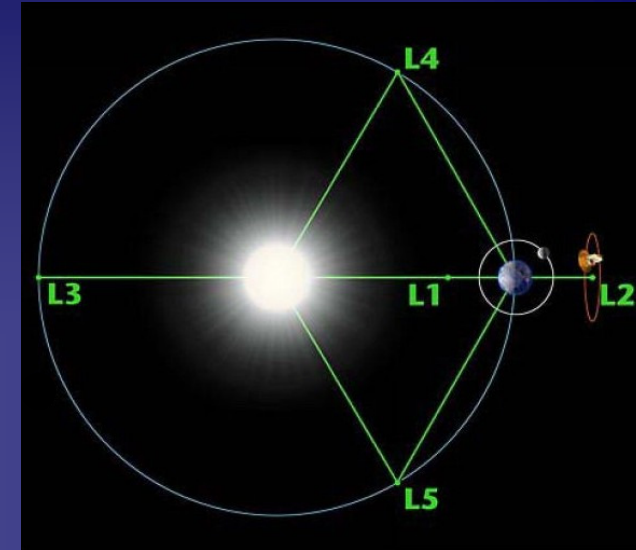
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PACS

Photodetector Array Camera and Spectrometer

Wavelength range: 57 - 210 μm

25x16 pixel (area: 49"x 49")

HERSCHEL - What do we need it for?

Herschel

Mirror: 3.5 m, $T \sim 80$ K
Launch: 2008
Orbit around L2
FIR space facility (57-670 μm)

Instruments: PACS, SPIRE, HIFI

Main Goals

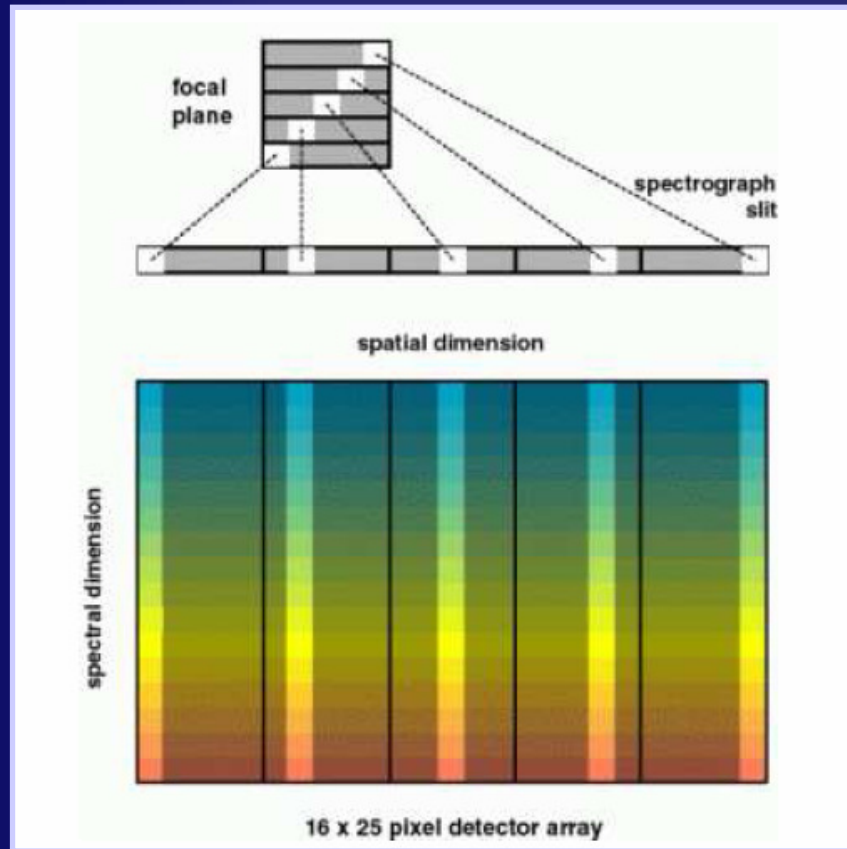
- ✘ Galaxy formation in the early universe
- ✘ Star formation & ISM
- ✘ Galaxies
- ✘ Solar System

PACS

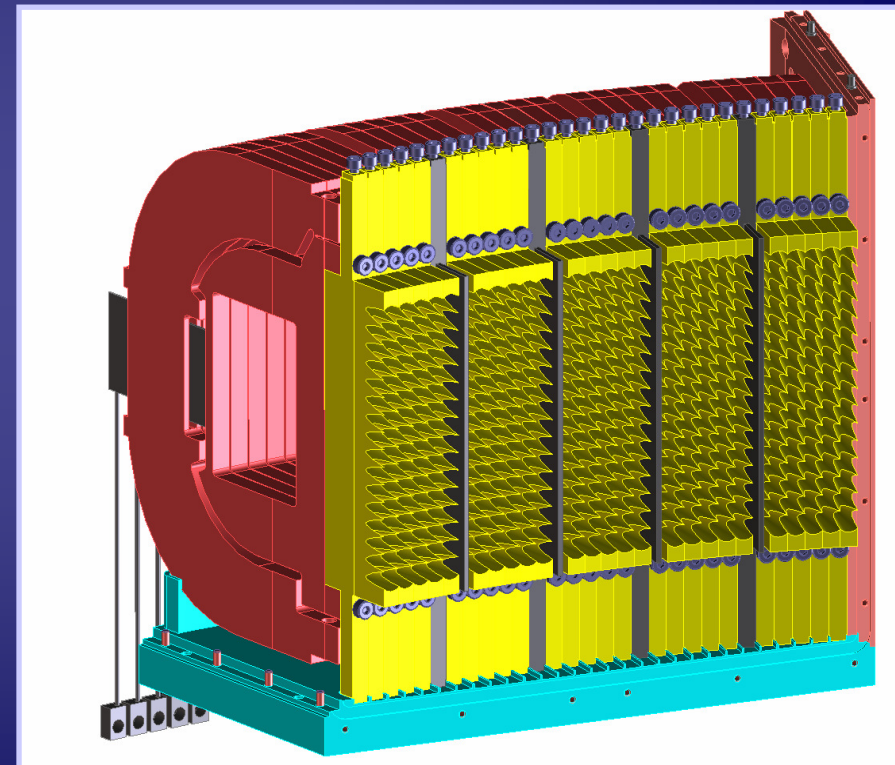
Photodetector Array Camera and Spectrometer

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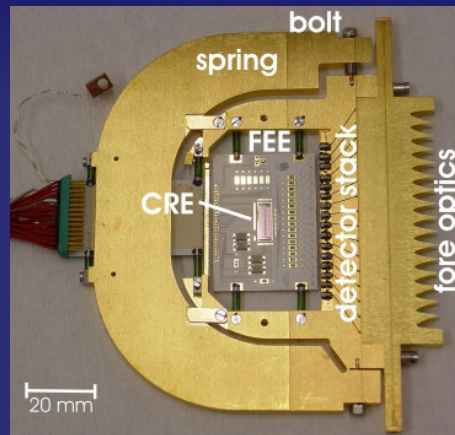
Spectrometer Camera



5x5 pixel (9.4" x 9.4" each)
⇒ **area: 49" x 49"**



FIR Photodetectors - Design



Ge:Ga crystals: stressed by

- spring
- bolt mechanism

Low-stressed (blue):

- 200N
- $\lambda_{\text{cutoff}} \sim 130 \mu\text{m}$

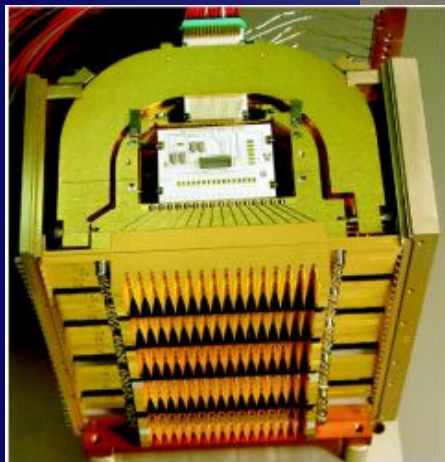
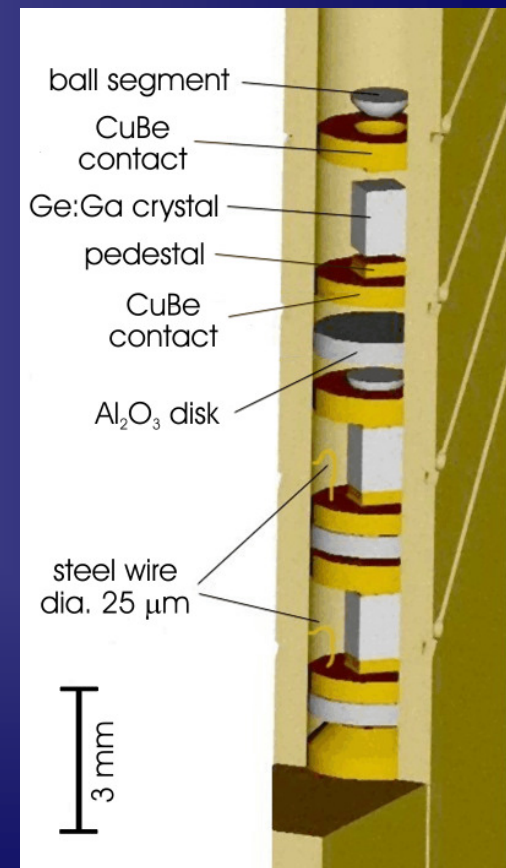
High-stressed (red):

- 800N
- $\lambda_{\text{cutoff}} \sim 210 \mu\text{m}$

Operation temperature:

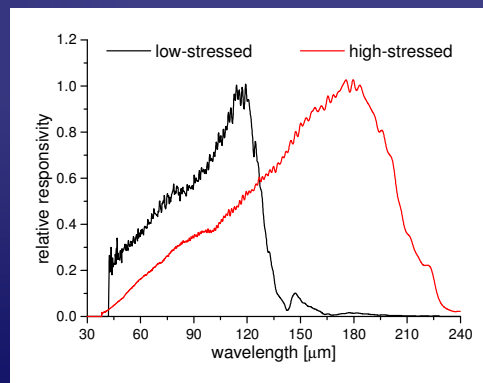
- $T_{\text{det}} \sim 1.8\text{K}, 2.5\text{K}$

Ge:Ga Detector stack



Detector module

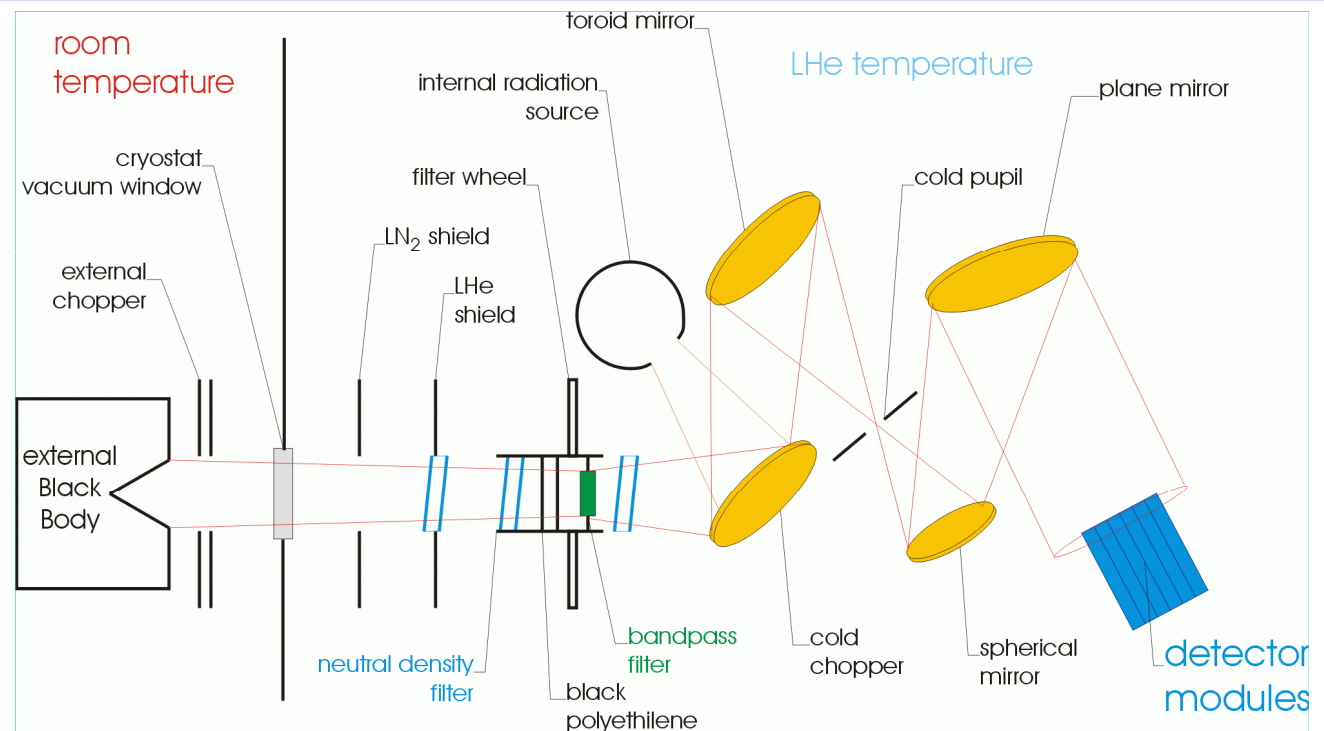
PACS:
16 x 25 pixel



Detector Test Setup

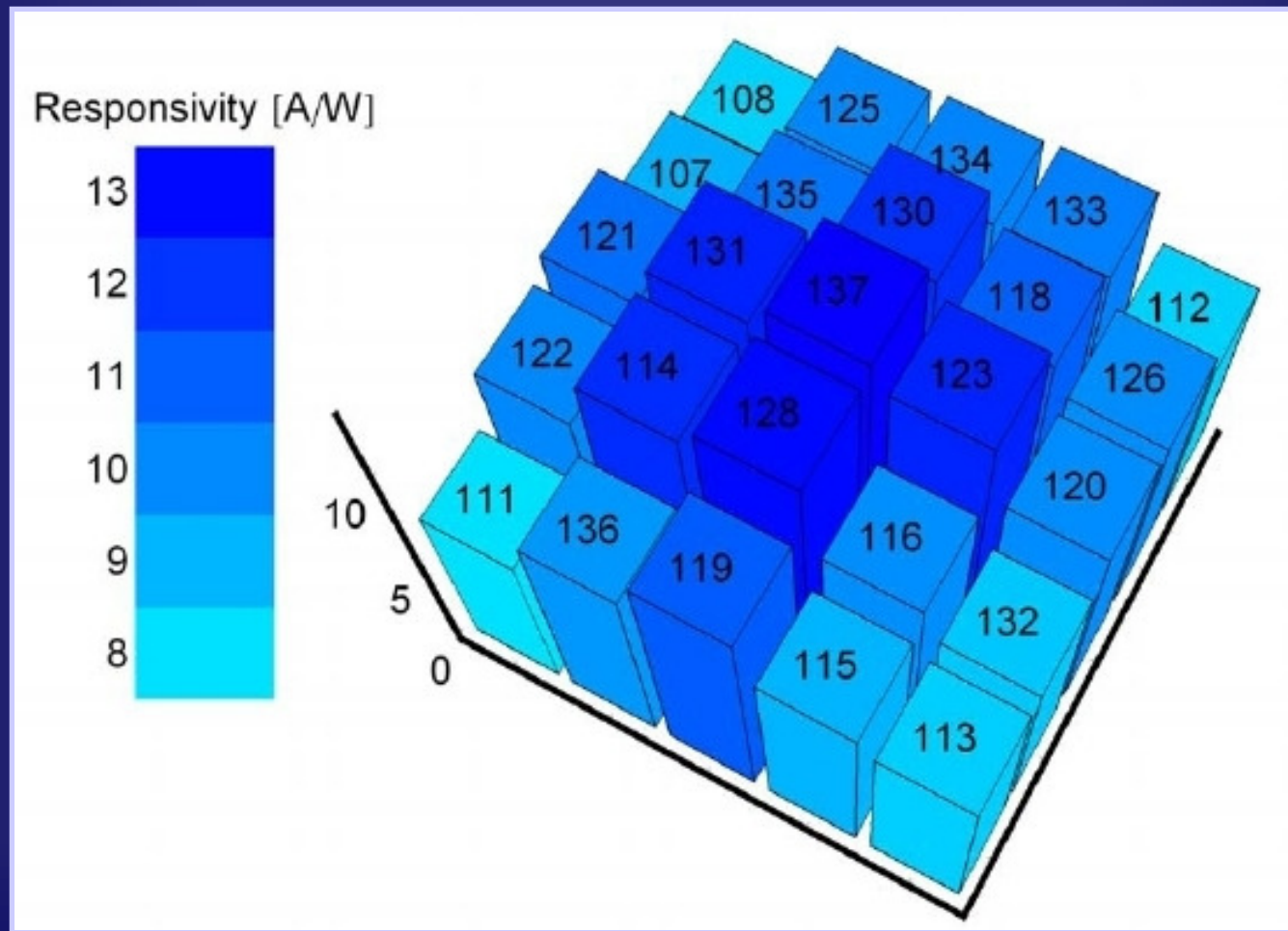


- Simulation of in-flight operation conditions
- Accurate IR fluxes



Characterisation

Responsivity: resulting detector current per incident IR flux power



Infrared Telescopes on the Moon - from a Robotic Precursor to a Very Large Telescope



To Moon and Beyond..

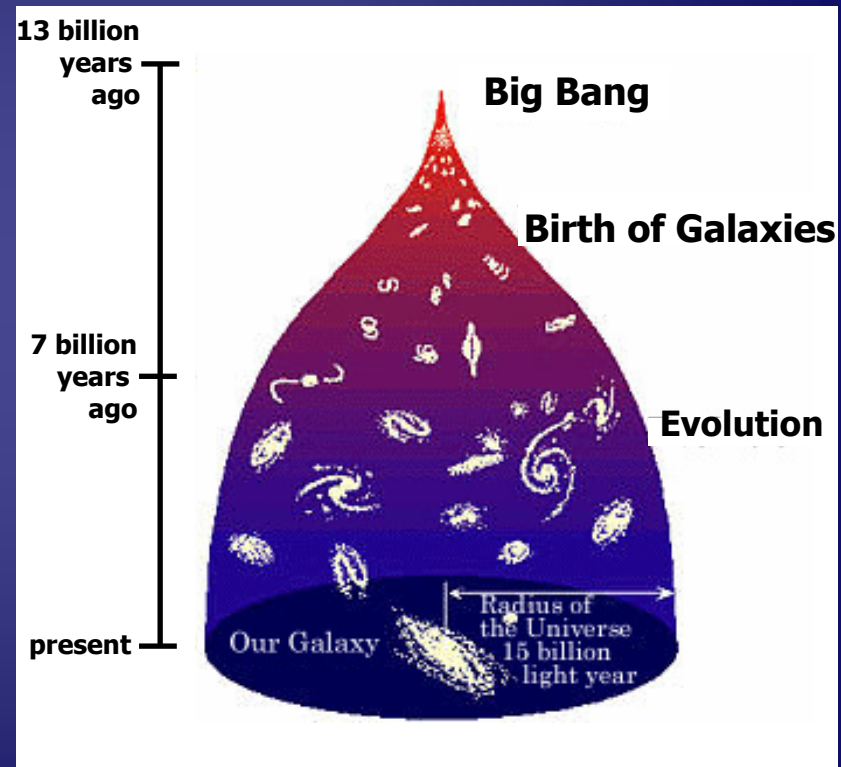
Today's Infrared Astronomy

Major scientific questions:

- First stars in the universe after BB
- Evolution of galaxies and quasars
- Cosmic Infrared Background
- Star birth and formation
- Extrasolar planets

Need for future:

- Higher sensitivity
- Very large aperture
- Larger detector arrays
- Ultra deep surveys



<http://www.ir.isas.jaxa.jp/ASTRO-F>

Infrared views of the dusty universe

Ground

Advantages:

- Large apertures
- Upgrade of instruments

Limits:

- Atmospheric windows
- Thermal background & airglow

Space

Advantages:

- All wavelength accessible
- Diffraction limited
- Passive cooling

Limits:

- High investment
- No service
- Apertures: 6.5 m

Opportunity
for future

Larger
telescopes
on moon?

Large liquid mirror telescope on the moon?

Moon

Advantages:

- Large stable platform
- No atmosphere
- Slow rotation
- Gravity
- Maintenance, upgrade

But:

- Manned flights stopped 30 years ago

Large liquid mirror telescope on the moon

Advantages:

- Gravity shapes perfect surface
- Large aperture for low costs
- Light weight structure

Limits:

- Zenith telescope

→ Ideal for deep observations

Status of liquid mirror telescopes

On ground (Borra et al.):

- Proven for 6m
- Liquid: Mercury
- Performance: Seeing limited

Challenges:

- Surface ripples
- Smooth & precise rotation



Large Zenith Telescope

Maple Ridge, BC

D=6m, f/1.5, CCD: 2Kx2K

Large IR lunar telescope

Basic concept

- Diameter > 20 m
- $1 < \lambda < 20\mu\text{m}$ (liquid mirror telescope)
- $20 < \lambda < 350\mu\text{m}$ (interferometer of solid mirrors)
- $T < 100\text{K}$
- Deployed & operated robotically

Technological challenges

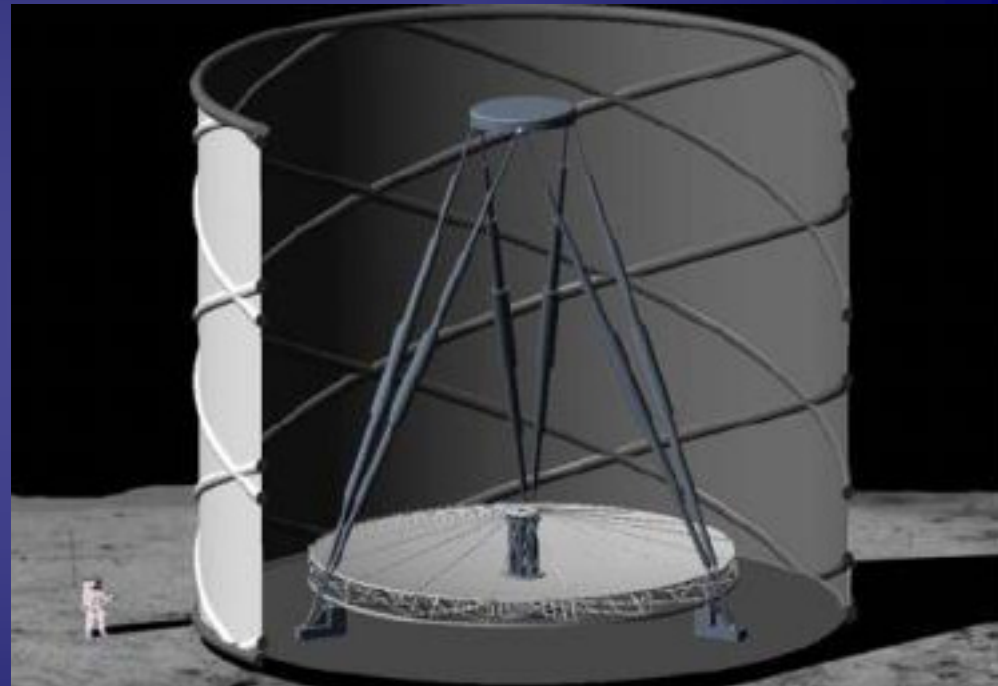
- Liquid
- Central bearing

Expected performance

- 3 x spatial resolution of JWST
- ~ 10 x higher sensitivity
- $F \sim 100\text{pJy}$

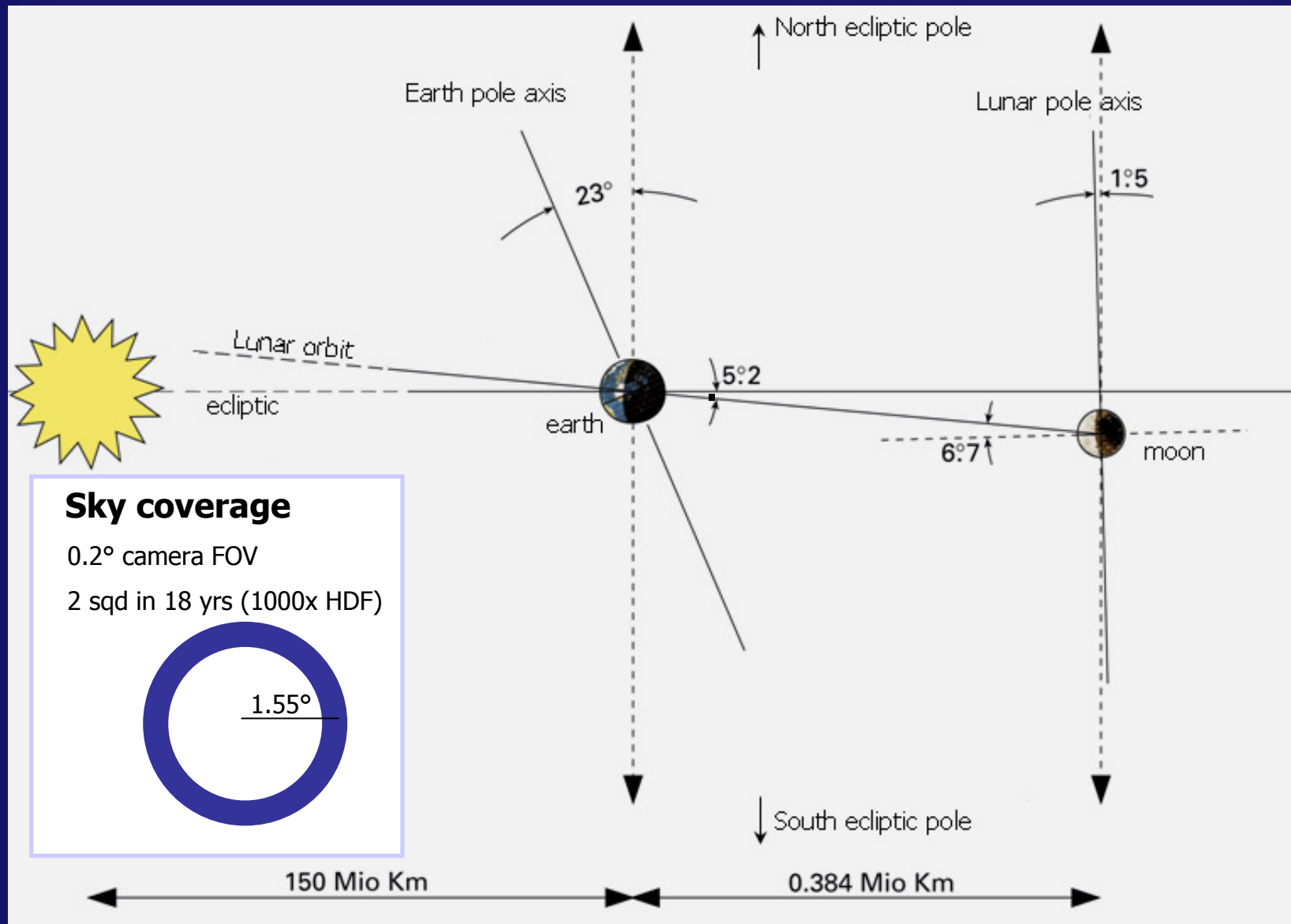
Operation

- Zenith pointing \rightarrow no steering, easy mounting
- Polar location preferred \rightarrow long exposures without tracking



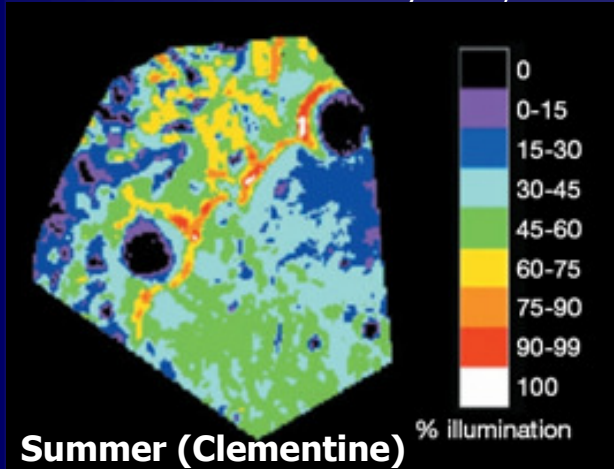
R. Angel, T. Connor (2004)

Lunar orbit



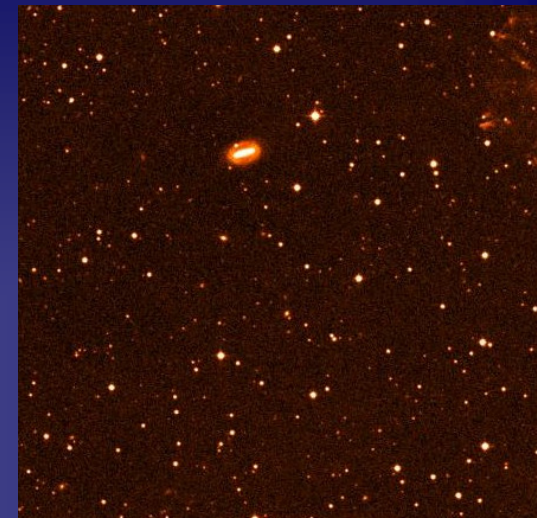
Pole selection for large lunar telescope

Bussey et al., 2005

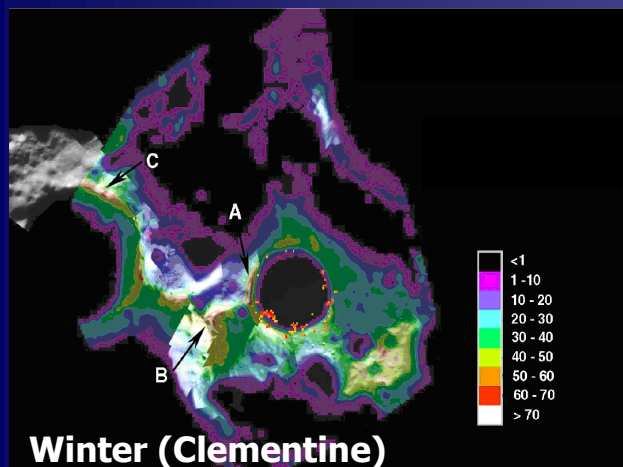


Peary crater (88.6°N)

- Permanent shadow
- Eternal light at rim

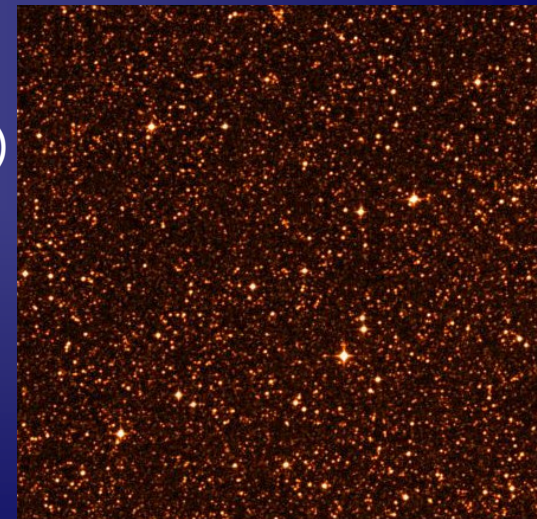


North Ecliptic Pole 15 x 15 arcmin



Shackleton crater (89.9°S)

- Permanent shadow
- Max. illumination 80-90%
- Large Magellanic Cloud



South Ecliptic Pole 15 x 15 arcmin

Is there need for a precursor mission ?

Is there dust?



Technological
challenges?

How dark is the sky?

Lunar environment

Nearly no atmosphere ($\sim 3 \times 10^{-15}$ bar) & no dipole magnetic field ($\sim 10^{-9}$ T)

- Faint atmospheric emission?
- $100 \text{ K} < T < 400 \text{ K}$
- High ionizing radiation background
- Impacts: meteorites & micrometeorites

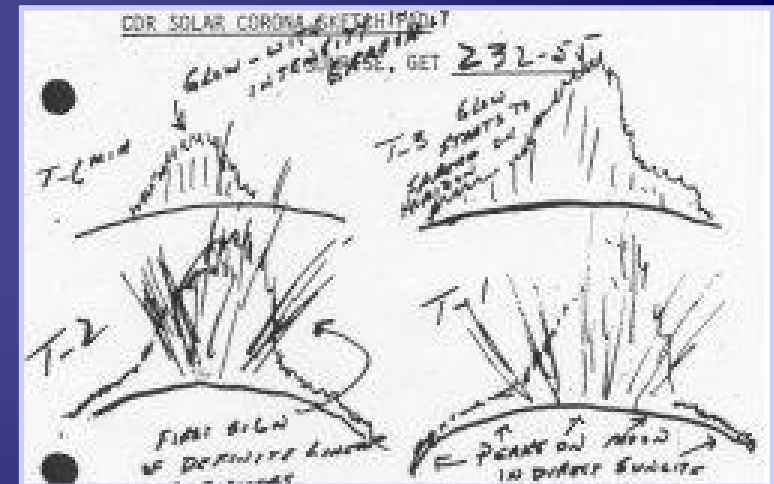
Dust

- Origin: meteorites
- Levitation near terminator (10-30cm)
- Thin dust atmosphere (>100km) ?



Need for precursor mission

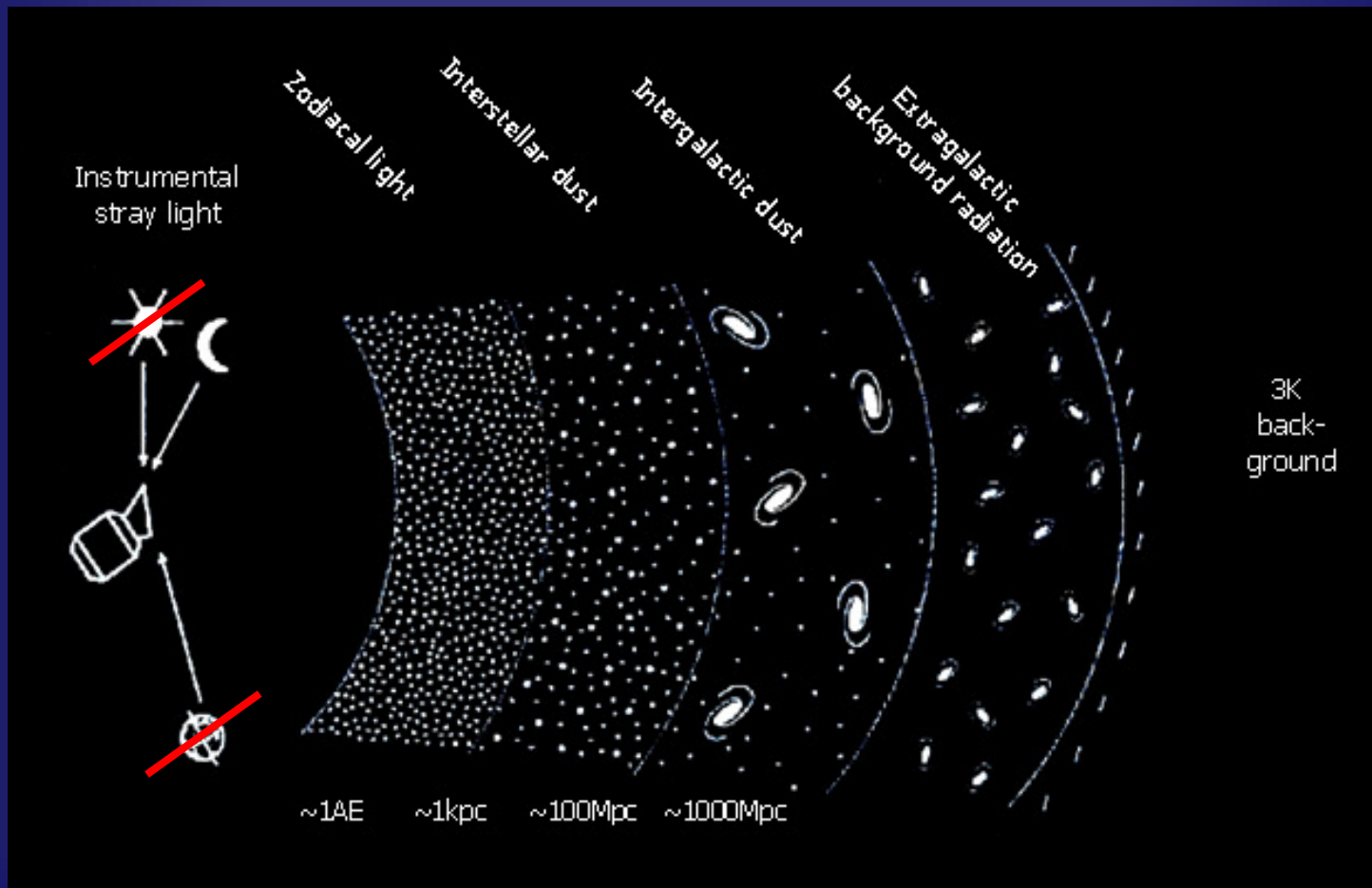
- Study cleanliness of lunar environment
- Additional scientific benefits



Lunar rays sketched by Apollo 17 captain Cernan (NASA)

Scientific benefit

- 1.) Absolute surface brightness of zodiacal light
- 2.) Measurements of EBL (constraints on cosmology)



Precursor design

First small instrument

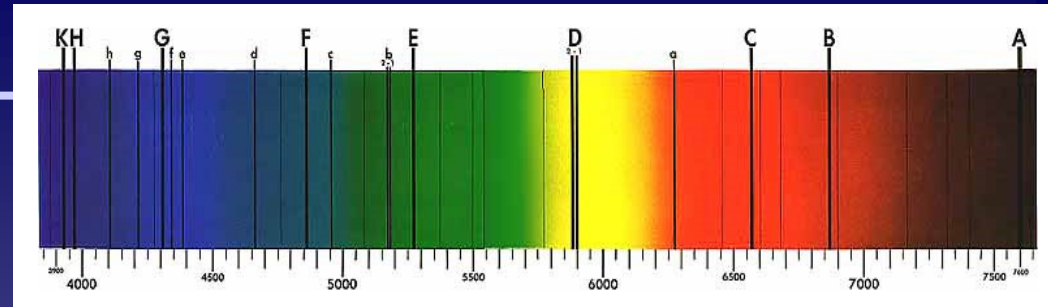
- Piggyback on lunar lander
- Wavelength: 0.4 – 2.5 μm
- FP or interference filter: $\lambda/\Delta\lambda \sim 1000$
- Prism
- Spatial resolution ≤ 1 arcmin, FoV $\sim 5^\circ$
- Elevation scans: 0-90 $^\circ$

Logistics

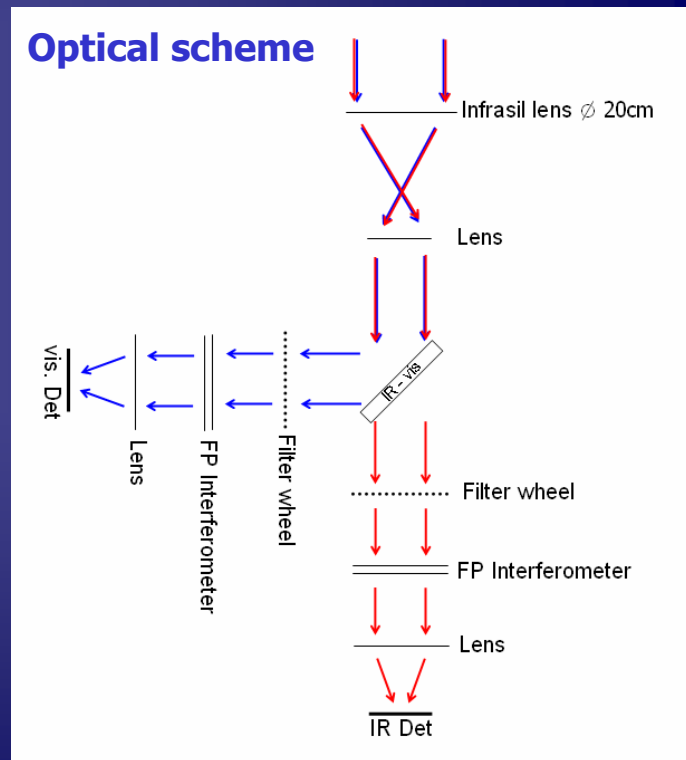
- Location of landing: back side, mid-latitude, pole,..
- Operation time: ≥ 14 days
- Cooling: passive cooling in crater or by radiation shields, new moon

Next

- Liquid mirror precursor: $\phi = 1.5\text{m}$ LMT



Solar Fraunhofer lines



Conclusion

Will the moon be a German?

„Wird der Mond ein Deutscher?“

(BILD)



Thanks for your interest !

